



Toolbox for the Modeling and Analysis of Thermodynamic Systems (T-MATS)

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Outline

- T-MATS Description
- Background
- Framework
- Block Sets
- Examples
- Conclusion
- Future work



T-MATS Description

- **Toolbox for the Modeling and Analysis of Thermodynamic systems, T-MATS**
 - Modular thermodynamic modeling framework
 - Designed for easy creation of custom Component Level Models (CLM)
 - Built in MATLAB®/Simulink®
- **Package highlights**
 - General thermodynamic simulation design framework
 - Variable input system solvers
 - Advanced turbo-machinery block sets
 - Control system block sets
- **Development being led by NASA Glenn Research Center**
 - Non-proprietary, free of export restrictions, and open source
 - Open collaboration environment



Background

- Thermodynamic simulation examples

Model Type	Examples
Steady-State (system convergence may be required)	Gas turbine cycle model <ul style="list-style-type: none">e.g., performance models
Dynamic with Quasi-steady-state variables (multi-iteration simulation; time and system convergence)	Gas turbine model with spool dynamics only. (real time running capability) <ul style="list-style-type: none">e.g., control models
Fully Defined Dynamic Simulation (iteration over time)	Dynamic gas turbine model with spool and volume dynamics (typically runs more slowly) <ul style="list-style-type: none">e.g., near stall performance models



Background: Industry Study

Package	User Friendly*	Flexibility*	Export Restricted	Source code available	Dynamic	Control System	Cost
C-MAPSS40k, NASA	High	Low	Yes	Yes	Yes	Yes	MATLAB
Matlab: ThermLib toolbox, Eutec	High	Medium	No	No	Yes	No	MATLAB + \$4900
Cantera, Open source	Low	High	No	Yes	No	No	None
Gas Turbine Simulation Program (GSP), NRL	Medium	Medium	No	No	Yes	Yes	\$4,000
GasTurb, Nrec	Medium	Low	No	No	Yes	Yes	\$1340
T-MATS, NASA	High	High	No	Yes	Yes	Yes	MATLAB

Definitions: 1* User Friendly, Controls Perspective

Low : Code based

Med: Model based

High: Model based with package implemented in a platform that is an industry standard

2* Flexibility

Low : Plant configuration set

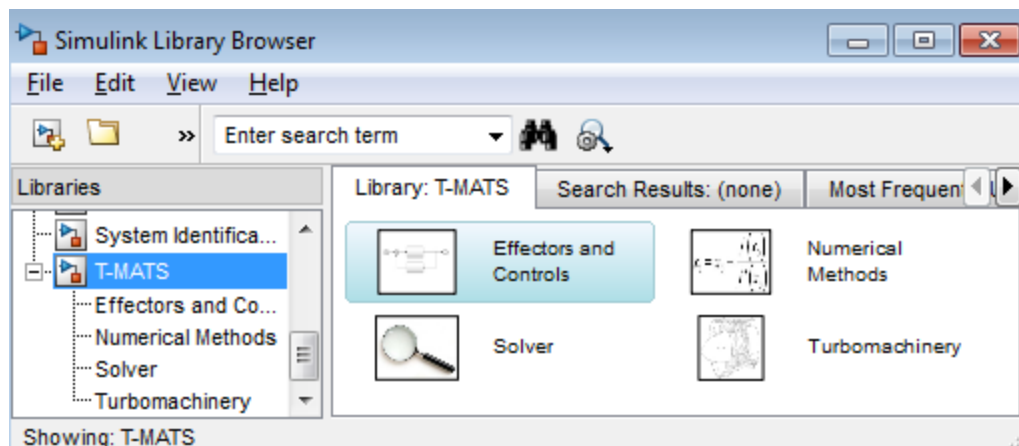
Med: Object oriented, objects difficult to update

High: Object oriented, objects easily adaptable by user



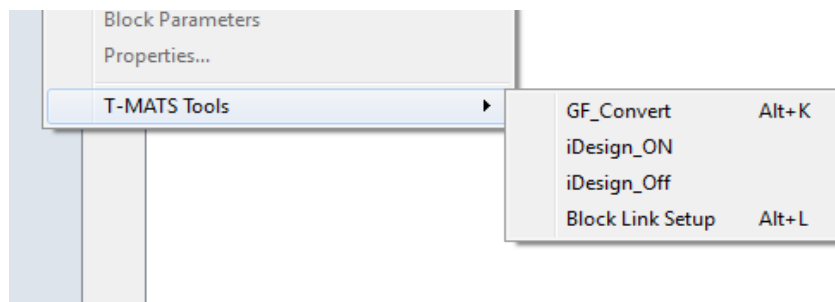
T-MATS Framework

- T-MATS is a plug-in for a MATLAB/Simulink platform
 - additional blocks in the Simulink Library Browser:



Added Simulink
Thermodynamic modeling
and numerical solving
functionality

- additional diagram tools for model development in Simulink:



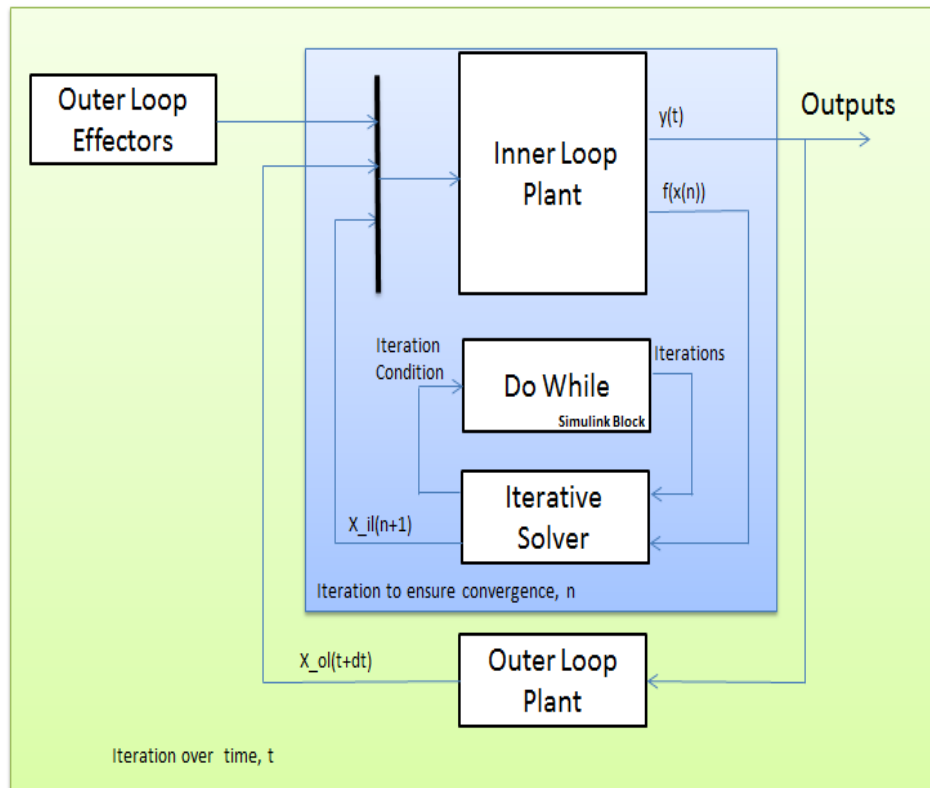
Faster and easier
model creation

T-MATS Framework

- Dynamic Simulation Example:

- Multi-loop structure

- The “outer” loop (green) iterates in the time domain
 - Not required for steady-state models
- The “inner” loop (blue) solves for plant convergence during each time step





Blocks: Numerical Solver

- Many Thermodynamic models are partially defined and require a solver to ensure model conservation (e.g., mass, energy, etc.).
 - In many gas turbine simulations, component flow will typically be solved by an independent solver.
- T-MATS contains solvers that perform in two main steps:
 - Automated Jacobian (system gradient) Calculation

$$J = \begin{bmatrix} \frac{\partial f_1}{\partial x_1} & \dots & \frac{\partial f_1}{\partial x_n} \\ \vdots & \ddots & \vdots \\ \frac{\partial f_m}{\partial x_1} & \dots & \frac{\partial f_m}{\partial x_n} \end{bmatrix}$$

- Each plant input is perturbed to find the effect on each plant output.
- Newton-Raphson method is used to “converge” the system.

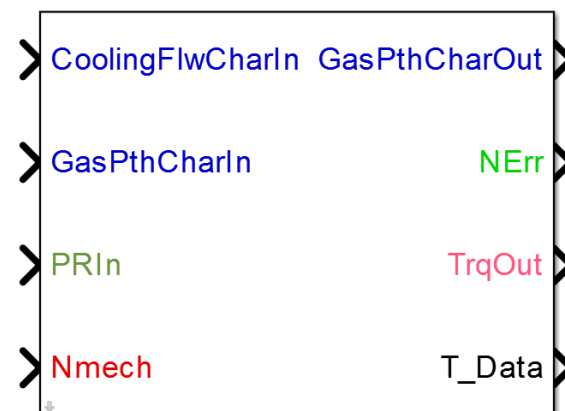
$$x(n+1) = x(n) - \frac{f(x(n))}{f'(x(n))} \quad \text{where,} \quad f'x(n) = J$$



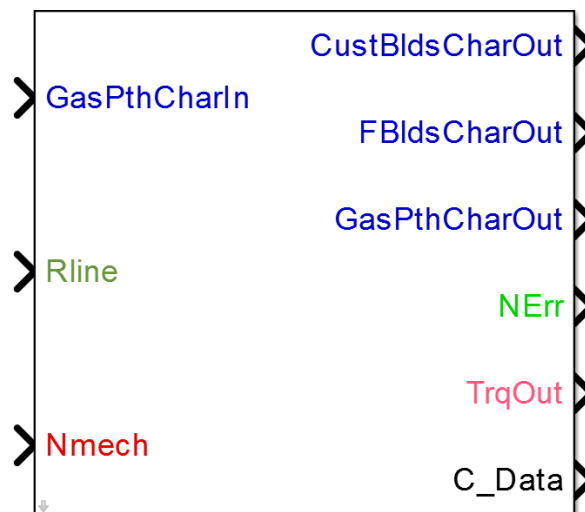
Blocks: Turbo-machinery

- T-MATS contains component blocks necessary for creation of turbo-machinery systems

- Models based on common industry practices
 - Energy balance modeling approach
 - R-line compressor maps in Compressor model
 - Pressure Ratio maps in Turbine model
 - Single fuel assumption
 - Flow errors generated by comparing component calculated flow with component input flow
- Includes blocks such as; compressor, turbine, nozzle, flow splitter, and valves among others.
- Built with S-functions, utilizing compiled MEX functions



Turbine



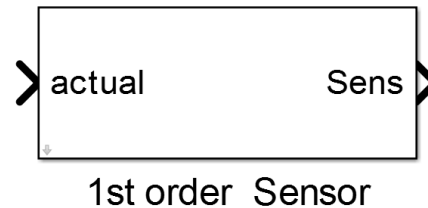
Compressor



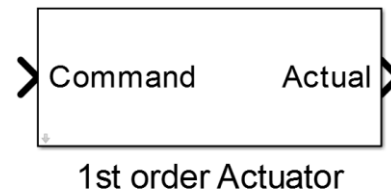
Blocks: Controls

- T-MATS contains component blocks designed for fast control systems creation

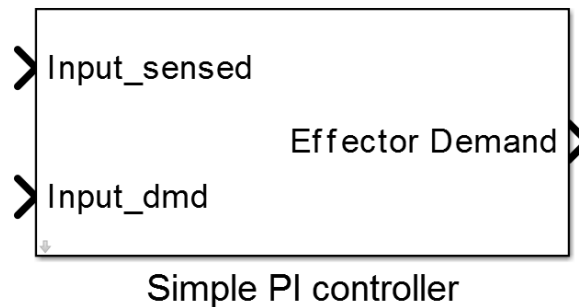
- Sensors:



- Actuators:



- PI controllers:





Blocks: Settings

- The T-MATS Simulation System is a highly tunable and flexible framework for Thermodynamic modeling.
 - T-MATS block Function Block Parameters
 - fast table and variable updates
 - Open source code
 - flexibility in component composition, as equations can be updated to meet system design
 - MATLAB/Simulink development environment
 - user-friendly, powerful, and versatile operation platform for model design

Function Block Parameters: Compressor

T-MATS: Compressor Library Block (mask) (link)

This block simulates the performance of a compressor using basic thermodynamic equations, properties, and table lookups.

C-Map | Bleeds | Stall Margin | iDesign

Y_C_NcVec_M - Compressor Map Corrected Speed Vector (Y-axis)

[0.500 0.900 1.050]

X_C_RlineVec_M - Compressor Map Rline Vector (X-axis)

[1.000 3.000]

T_C_Map_WcArray_M - Compressor Map Flow Array ($W_c = f(N_c, R_{line})$)

[0 0; 0 0; 0 0]

T_C_Map_PRRArray_M - Compressor Map Pressure Ratio Array ($PR = f(N_c, R_{line})$)

[0 0; 0 0; 0 0]

T_C_Map_EffArray_M - Compressor Map Efficiency Array ($Eff = f(N_c, R_{line})$)

[0 0; 0 0; 0 0]

s_C_Nc_M - Corrected Speed Scalar Constant (C_{Nc})

0.0001

s_C_Wc_M - Flow Scalar Constant (C_{Wc})

0.4953

s_C_PR_M - Pressure Ratio Scalar Constant (C_{PR})

0.8636

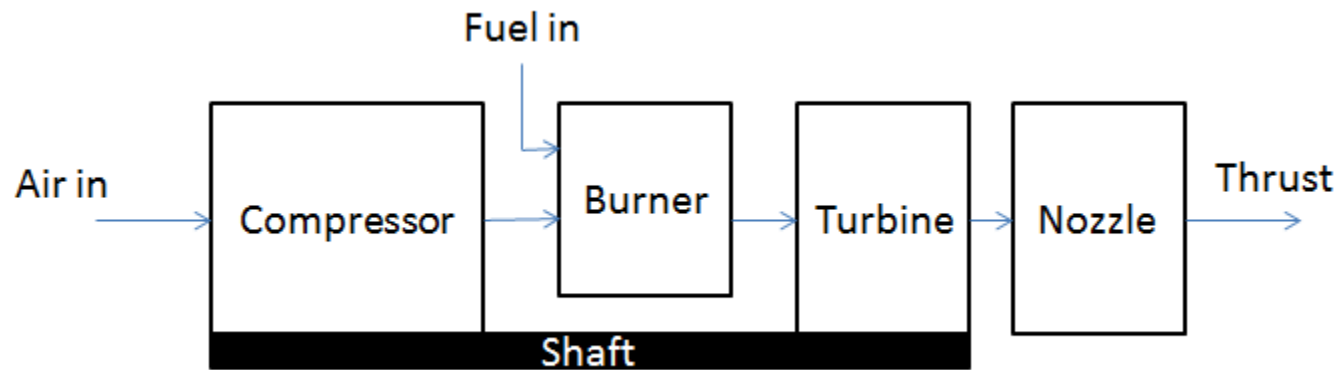
s_C_Eff_M - Efficiency Scalar Constant (C_{Eff})

0.9977

OK Cancel Help Apply

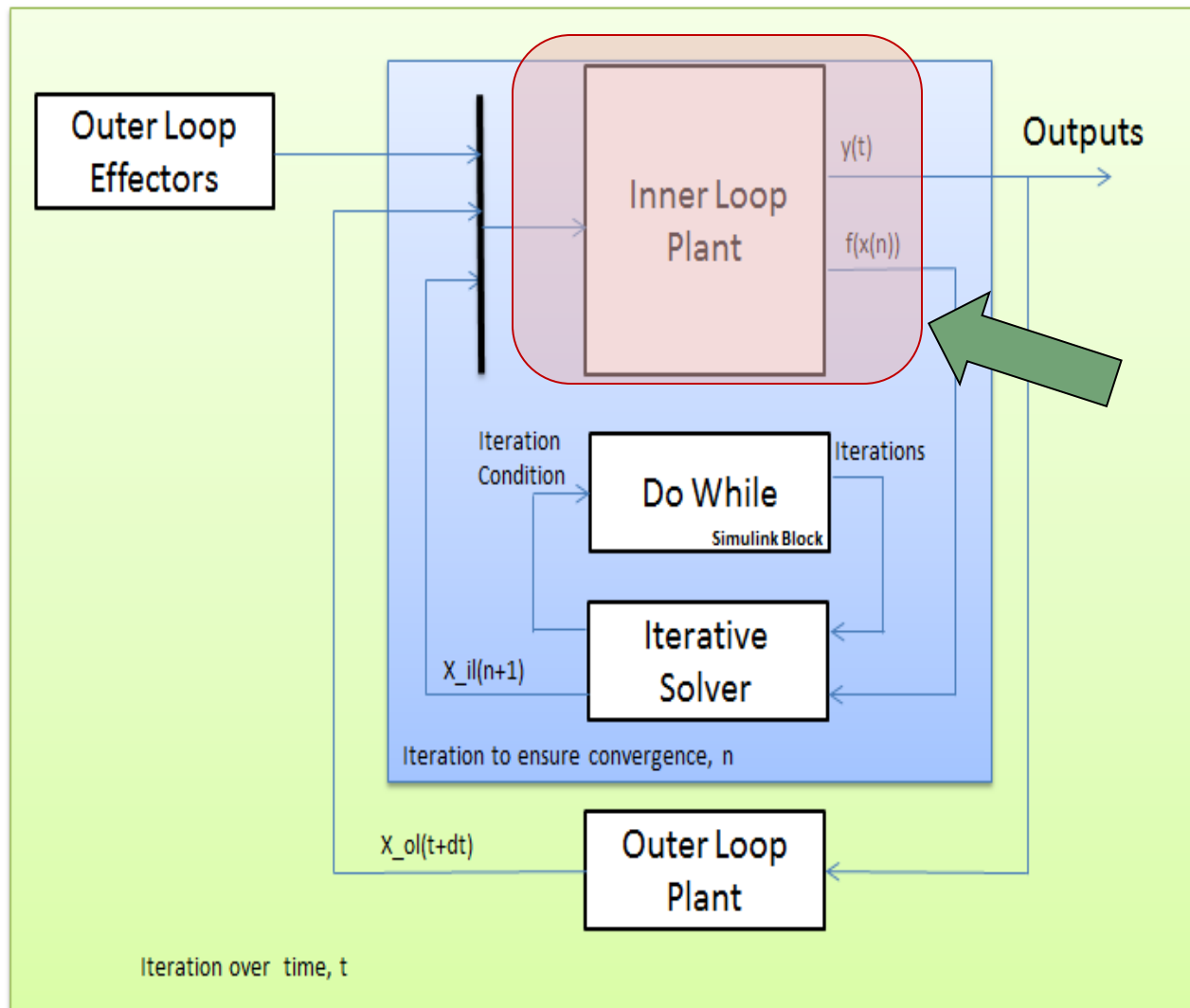


Dynamic Gas Turbine Example: Objective System

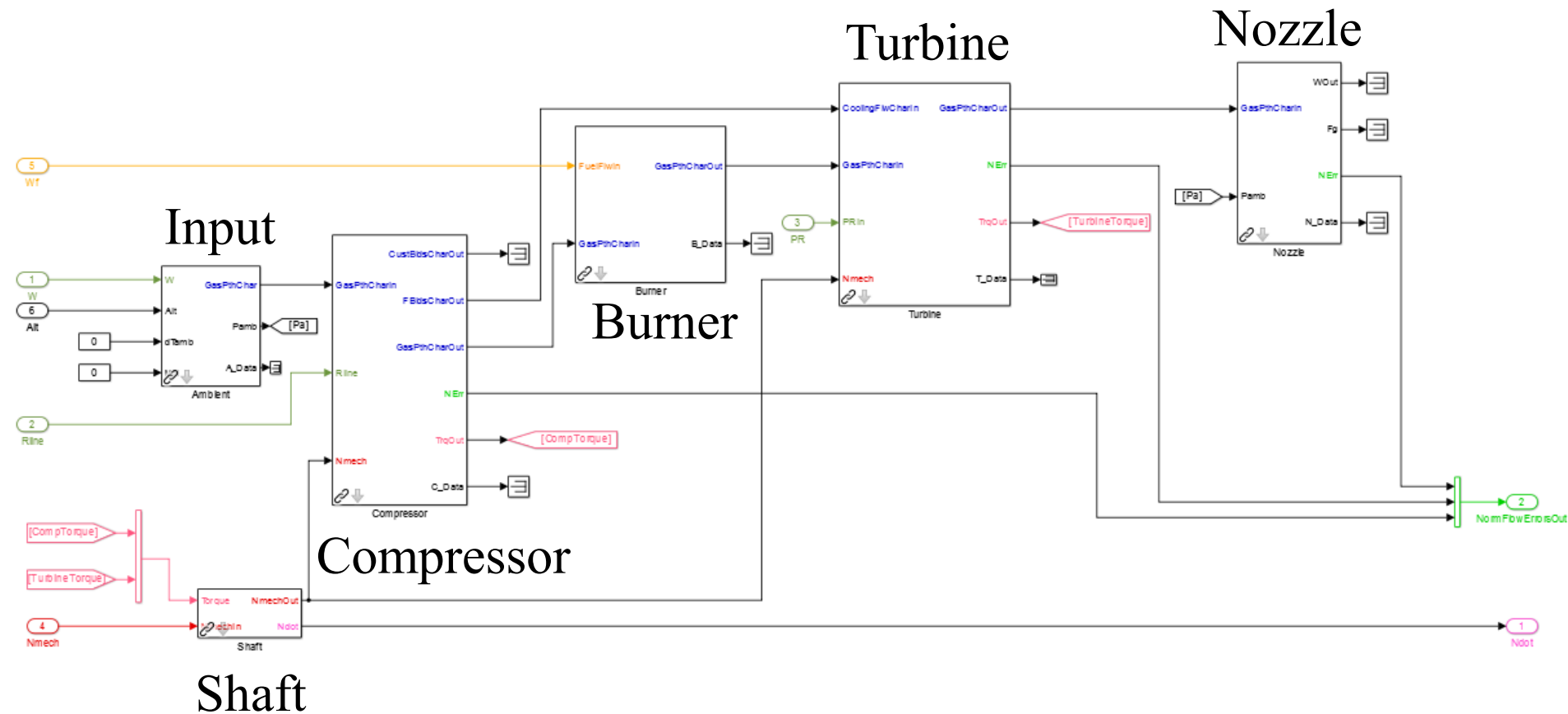


Simple Turbojet

Dynamic Gas Turbine Example: Creating the Inner Loop

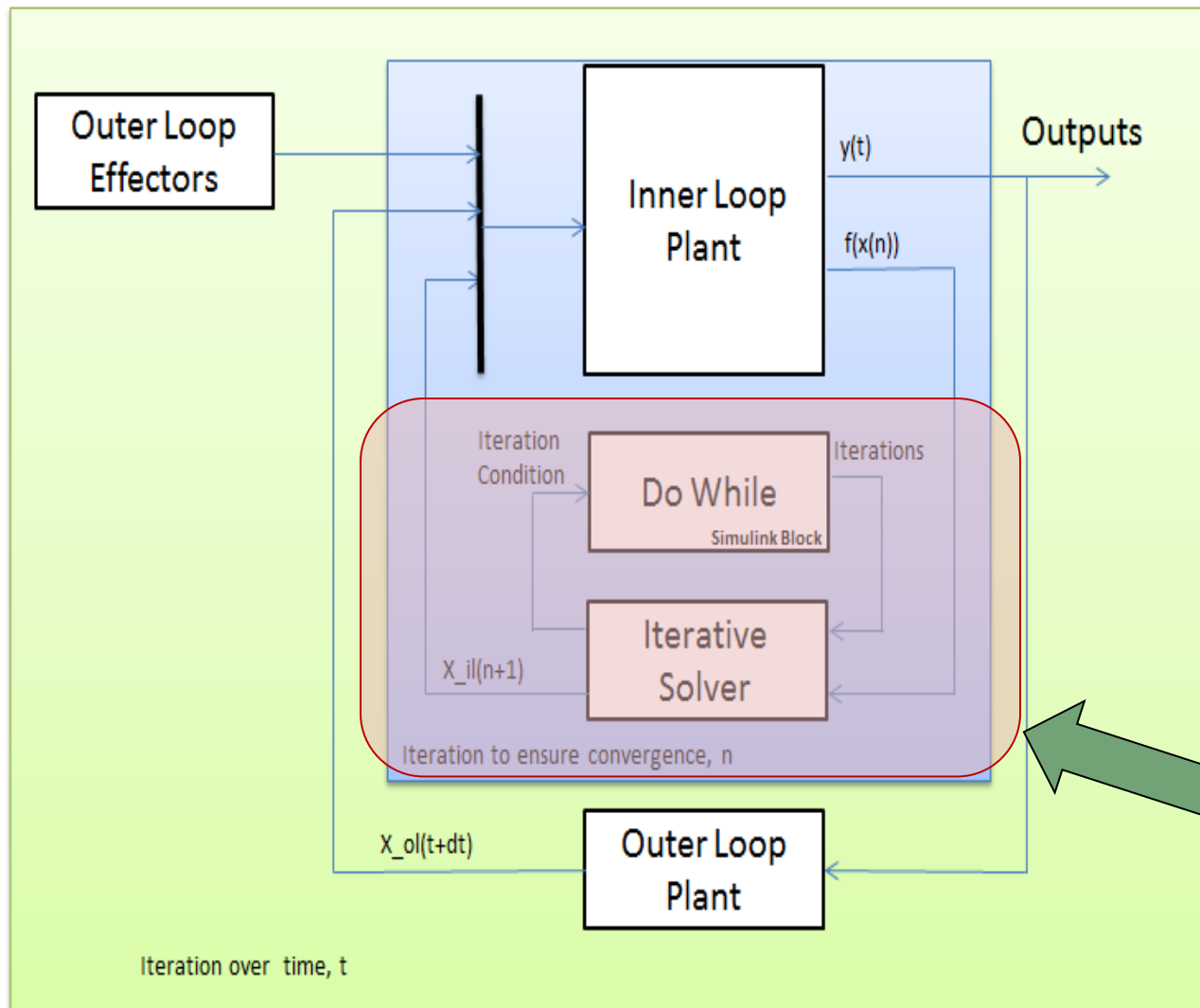


Dynamic Gas Turbine Example: Inner Loop Plant

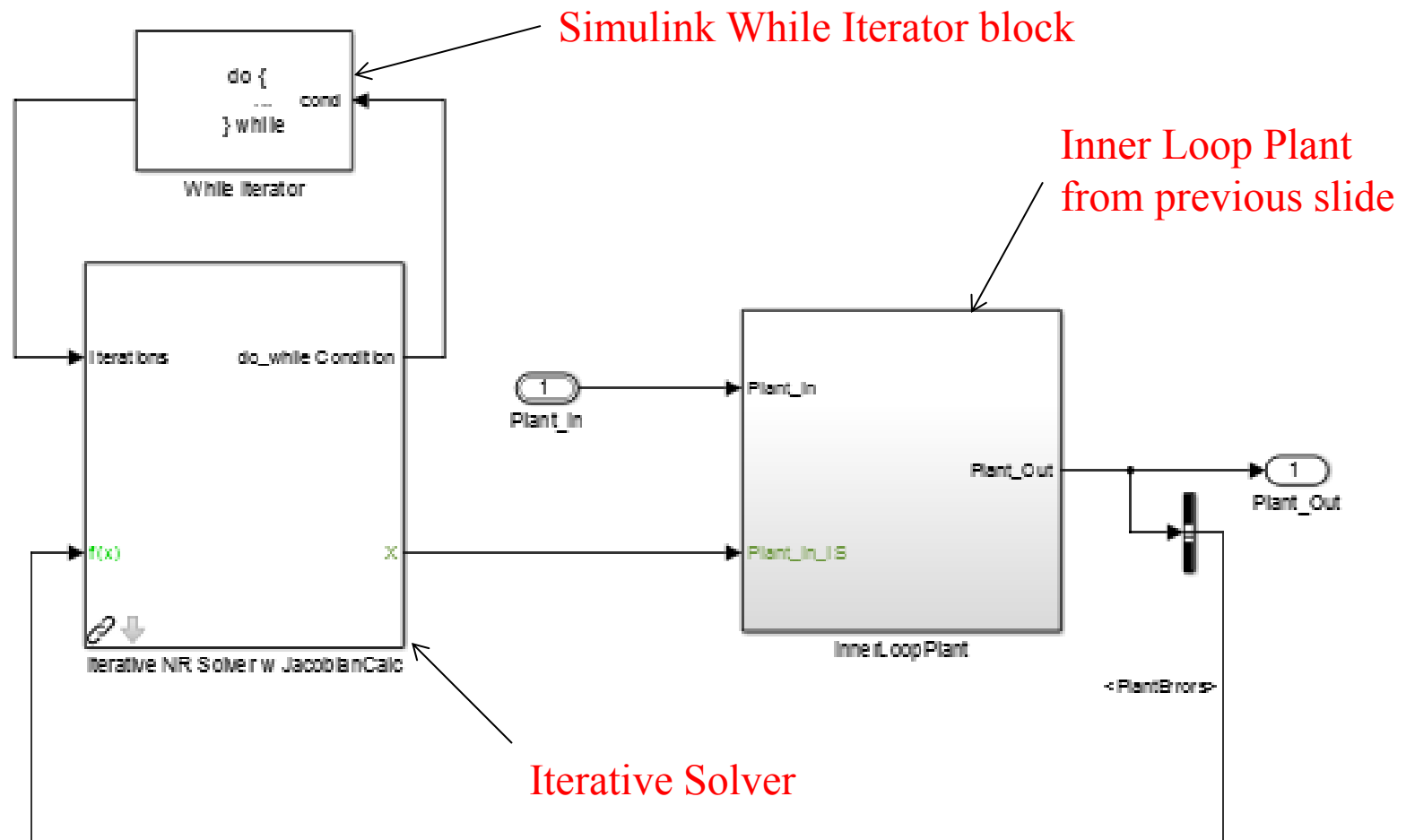


Turbojet plant model architecture made simple by T-MATS vectored I/O and block labeling

Dynamic Gas Turbine Example: Creating the Solver

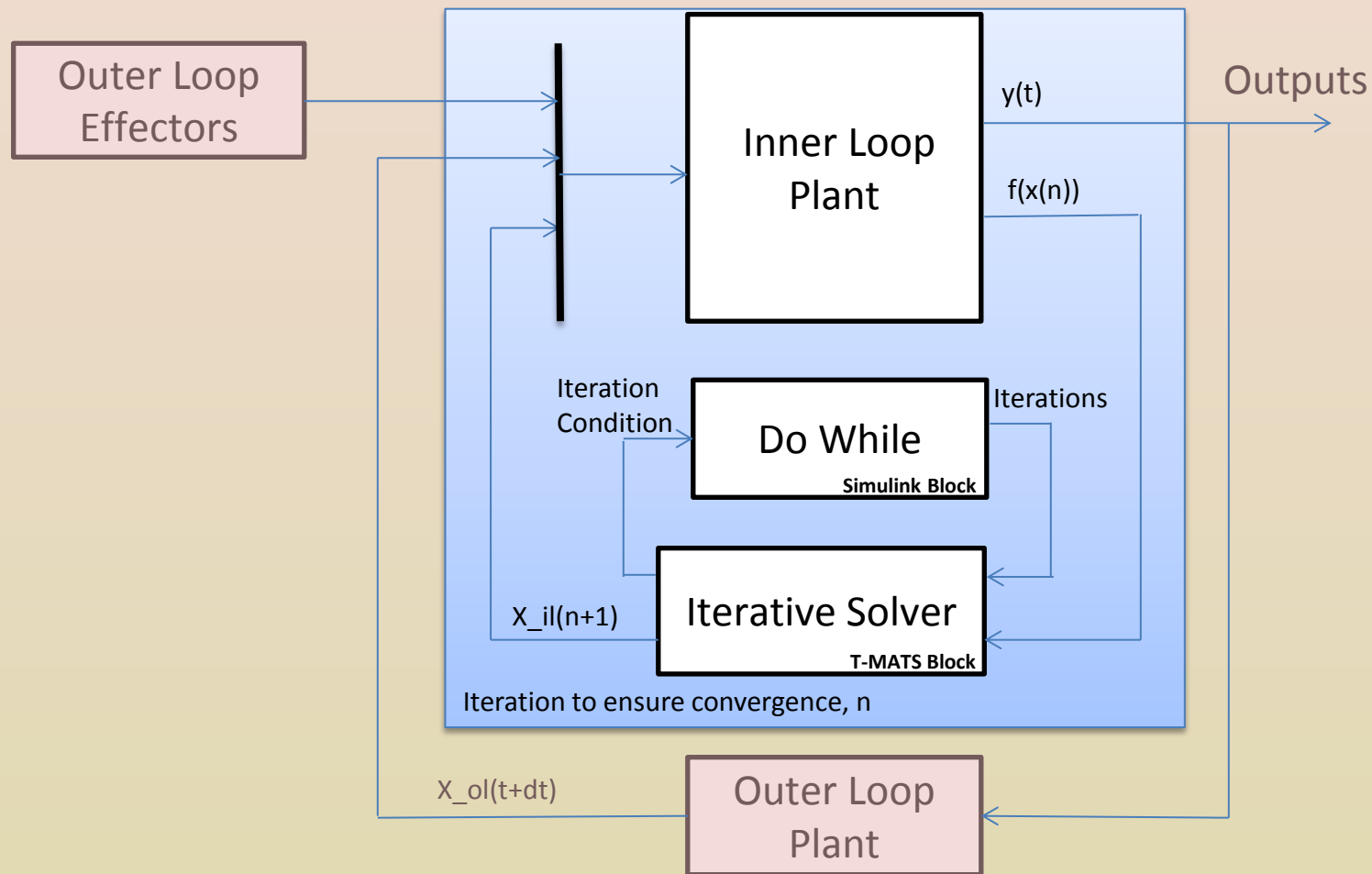
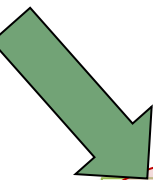


Dynamic Gas Turbine Example: Solver



Plant flow errors driven to zero by iterative solver block in parallel with While Iterator

Dynamic Gas Turbine Example: Creating the Outer Loop

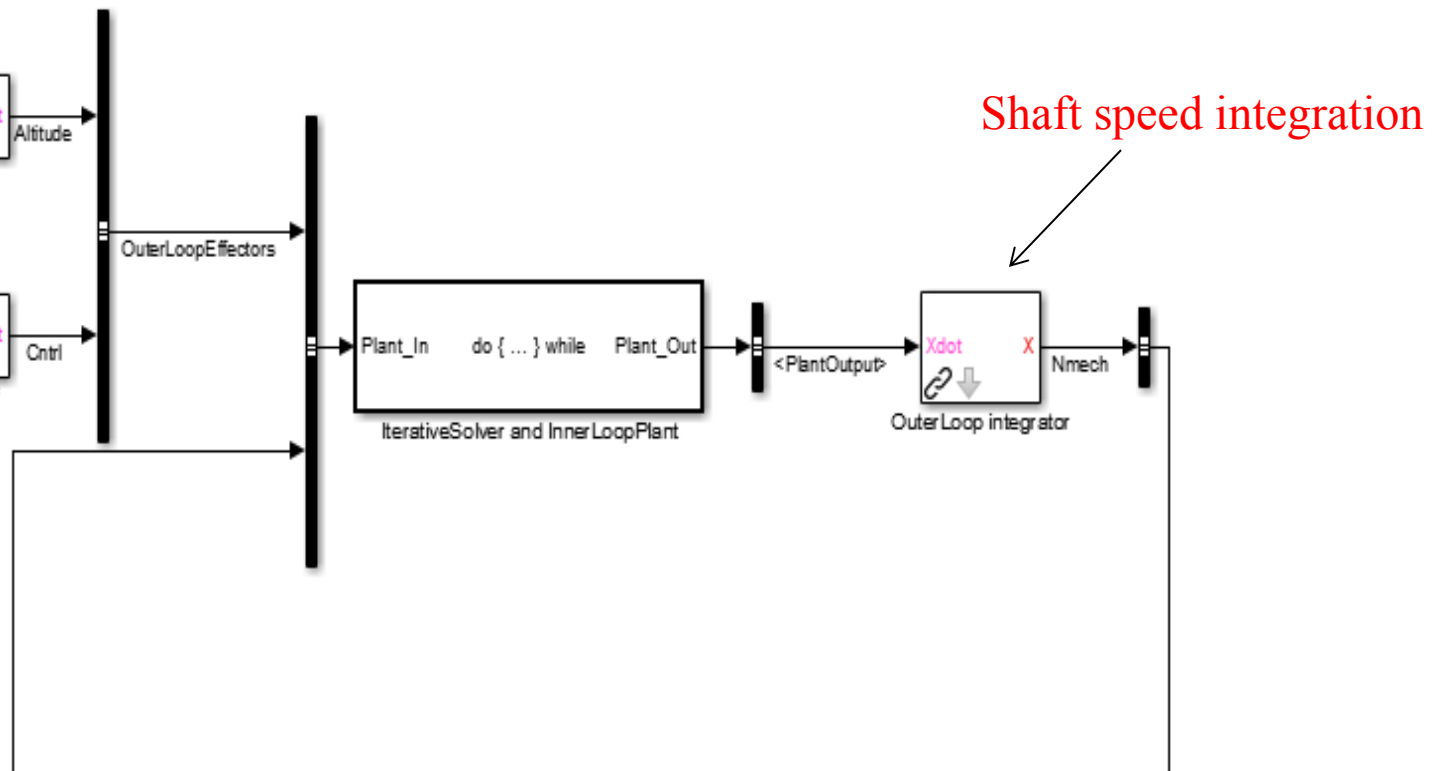


Iteration over time, t

Dynamic Gas Turbine Example: Outer Loop Plant

Environmental
conditions

Simple Control
System



Shaft integrator and other Outer Loop effectors added to create full system simulation



Verification and Release

- Verification was performed by matching T-MATS simulation data with other established simulations.
 - Models chosen for verification
 - NPSS steady-state turbojet model
 - C-MAPSS – High bypass turbofan engine model
 - In all cases differences in simulation performance were within acceptable limits.
- Expected Release: Q4,2013 or Q1,2014.
 - Pre-built examples will include:
 - Newton-Raphson equation solver
 - Steady state turbojet simulation
 - Dynamic turbojet simulation



Conclusions

- T-MATS offers a comprehensive thermodynamic simulation system
 - Thermodynamic system modeling framework
 - Automated system “convergence”
 - Advanced turbo-machinery modeling capability
 - Fast controller creation block set



Future Work

- **Increase thermodynamic modeling capability**
 - Introduce Cantera to T-MATS
 - “Cantera is a suite of object-oriented software tools for problems involving chemical kinetics, thermodynamics, and/or transport processes”
 - Open source
 - Increases thermodynamic modeling capability to include:
 - **Non-fuel specific gas turbine modeling**
 - **Fuel cells**
 - **Combustion**
 - **Chemical Equilibriums**

